

ILLINOIS OBSERVING NANOSATELLITE (ION)

**University of Illinois
Urbana-Champaign**

ILLINOIS TINY-SAT INITIATIVE (ITSI)

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OVERVIEW

How ION

- Educates and develops interest in space engineering
 - Brings flight experience to the space technology development community
 - Performs a focused earth science exploration mission
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ION MISSION

■ Education

- Provide a multidisciplinary engineering experience
- Give students familiarity with project management
- Develop interest in space engineering

ION MISSION

■ Technology

- Primary
 - Space qualification of vacuum arc micro-thruster propulsion system for Alameda Applied Sciences Corporation
 - Space qualification of Tether Applications Inc. Small Integrated Datalogger (SID) single-board computer
 - Development of an active, 3-axis magnetic attitude control system
- Secondary
 - Demonstration of on-board CMOS camera capabilities

■ Science

- Investigation of airglow layer of Earth's atmosphere through use of an on-board sensor [Photo-Multiplier Tube (PMT)]

Educational Mission

Multidisciplinary design experience

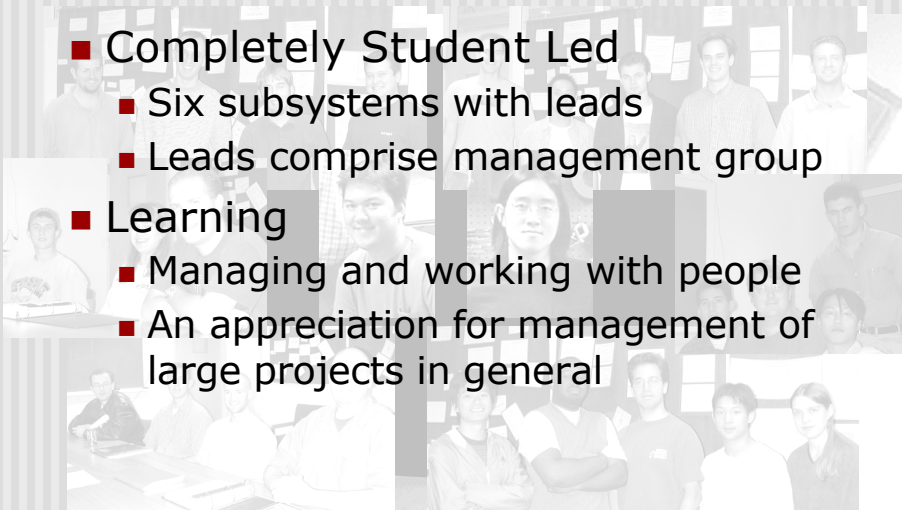
- Involves three departments
 - Aeronautical and Astronautical Engineering
 - Electrical and Computer Engineering
 - Computer Science
- Learning
 - How to coordinate efforts amongst a large and diverse group
 - Communication across discipline lines
 - An appreciation for other engineers' ideas
 - Versatility and flexibility



Educational Mission

Project Management

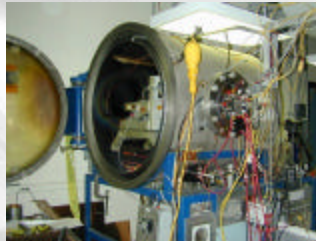
- Completely Student Led
 - Six subsystems with leads
 - Leads comprise management group
- Learning
 - Managing and working with people
 - An appreciation for management of large projects in general



Educational Mission

Develop interest in space engineering

- Excitement of putting something into space
- Introduction with the challenges involved
- Contact with the space community



Technology Mission

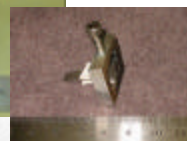
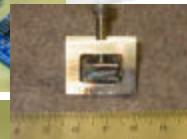
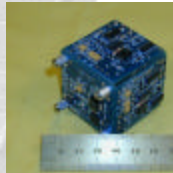
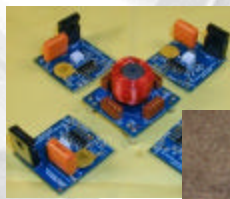
Space qualify VApT



Alameda Applied Sciences Corporation

Pulsed Electric "Vacuum Arc micro-Thruster"

- Uses electrical energy to accelerate inert solid fuel (Aluminum)
- Pulsed operation allows various thrust and power
- Very low mass



Technology Mission

Space qualify VA μ T

■ Thruster heads

- Cylindrical or “BLT”
- BLT design chosen due to higher reliability
- Current design uses satellite frame for propellant in a **self-consuming** fashion

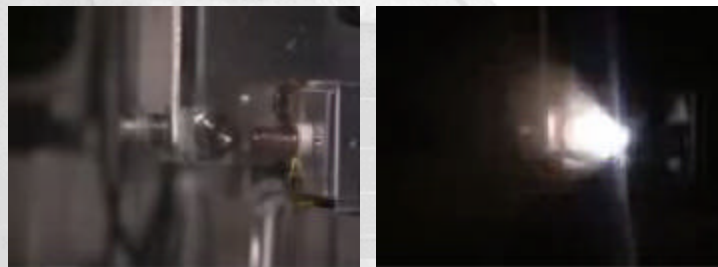


Technology Mission

Space qualify VA μ T

■ Utility

- Attitude Control
- Orbit Maintenance
- Orbit Transfer



Technology Mission

Space qualify VApT

■ Verification Experiments

■ Current Sensor

- Sensor to detect flow of current to thruster head
- Indicates if power is being supplied correctly

■ Attitude Determination

- Measure any changes in attitude caused by use of propulsion system

■ Propellant Deposition Monitor

- Ceramic wafer lightly coated with metal
- Back-splash from thrust head accumulates on wafer
- Resistivity level monitored

Technology Mission

Space qualify SID



■ Small Intelligent Datalogger (SID) by Tether Applications used for flight computer

■ Features

- Hitachi SH7045F 32-bit RISC microcontroller
- Low power: 15 MHz at < 0.5 W
- 1.2 MB RAM, 8 MB Data Storage



Technology Mission

Space qualify SID

■ On-board facilities

- Low-grade gyros and accelerometers
 - Confirmation of attitude information
- Redundant alarm clocks
 - Keep time
 - Scheduling
- Temperature sensor interface
 - Can measure temperatures in over 60 locations throughout the satellite
- CMOS camera interface
 - Collect images from the PhotoBit PB-300 based camera

Technology Mission

Space qualify SID

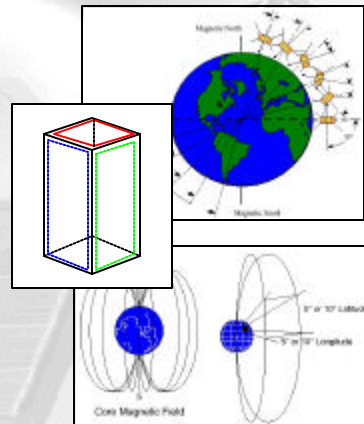
■ Additional resources

- 10-bit A/D, multiplexed many ways
 - Read solar array voltages, current sensors, etc.
- Serial peripheral
 - Communicate with communication system
- SID Rad-hard feature
 - Overcurrent protection circuitry
 - Power cycles the computer upon unexpected current increase
- Software development
 - Mostly in C/C++
 - Using free and familiar GNU development tools

Technology Mission

Develop 3-axis attitude control system

- Actuators
 - Free air torque coils
- Sensors
 - Primary
 - 3-Axis Magnetometer
 - Secondary
 - Rate Gyros
 - Accelerometers
 - Solar panels
- Software
 - Readings referenced against model of magnetosphere for attitude determination



Technology Mission

Demonstrate CMOS camera

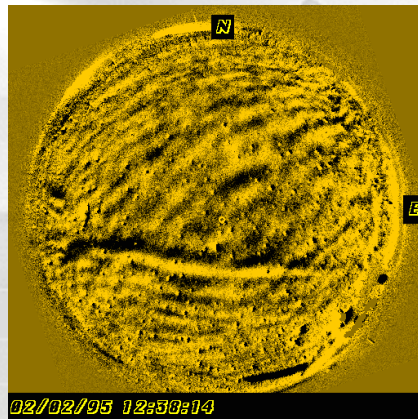


- Equipment
 - Photobit PB-300 Camera
 - SID camera interface
- Purpose
 - Additional tool for attitude determination
 - Pictures from our satellite will make for great PR

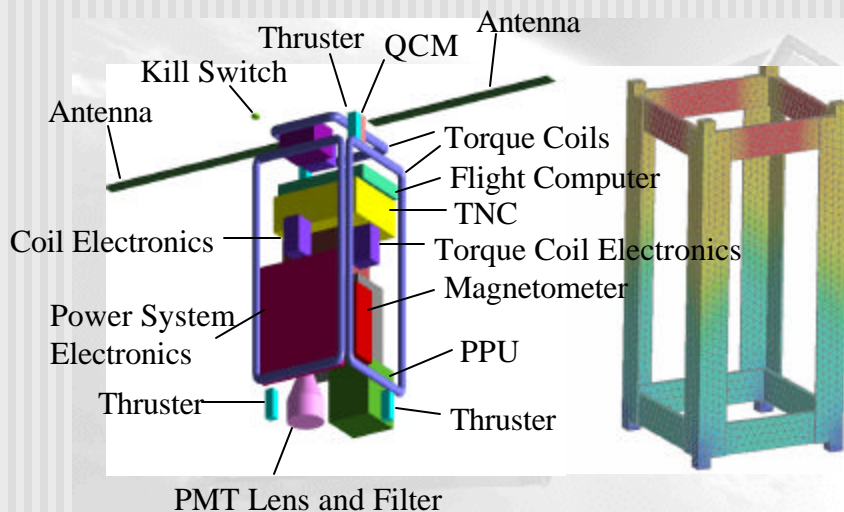
Science Mission

Investigate Air Glow Layer

- Image Air Glow Layer at 760nm
 - Study wave structures
 - Unable to image from ground due to absorption
- Equipment
 - Photomultiplier tube using photon counting technique

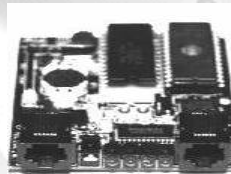


Support Structure



Support Communications

- Modified COTS parts
- 2W @ ~437 MHz in amateur radio band
- 1200 bps AFSK packet (AX.25)
- Half duplex
- Expected 150 KB/day with one ground station
- Two monopoles phased 180 deg apart



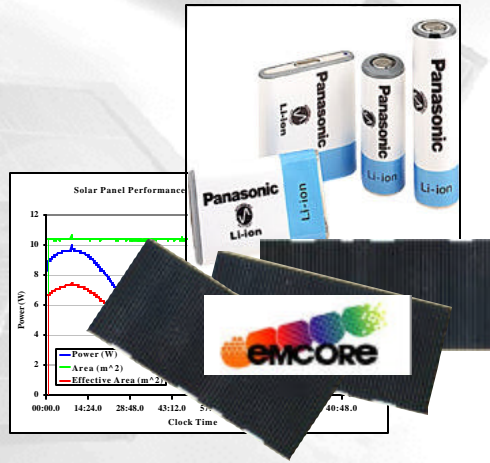
Support Ground Station

- Amateur radio satellite ground station
- 440, 1296, and 2400 MHz capable
- Automation for optimal use of short pass times



Support Power

- Emcore 26% efficient cells
- Panasonic Li-ion batteries
- Power use will be carefully scheduled
- Expect 7Wh/orbit from solar cells



Summary

- Education
 - Exposes budding engineers to space engineering
 - Project management experience
 - Interdisciplinary interaction
- Flight experience for space technology developers
 - Alameda Applied Sciences VAP_T propulsion system
 - Tether Applications Small Integrated Datalogger
 - Many commercial parts
- Technology
 - 3-axis control
- Science
 - Airglow layer study